



1.0 INPUT

1.1 Design Options

Design Code

American Standard LRFD

Unit

English Unit

Code

ANSI/AISC 360-16

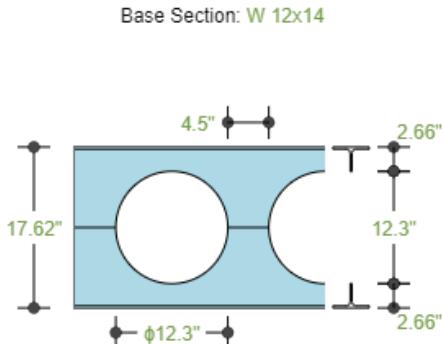
1.2 Section and Span

1.2.1 Base Section

Profile	Type = Standard
Section	= W 12x14
Overall Depth	D = 12.0 in
Overall Height	H = 11.9 in
Width of Flange	B = 3.97 in
Thickness of Flange	T = 0.23 in
Thickness of Web	t = 0.2 in

1.2.2 Span

Shape	= Straight
Beam Span	BL = 40 ft

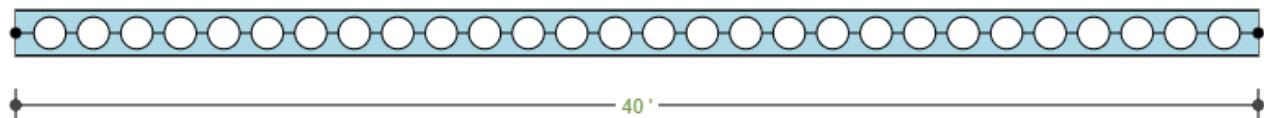


1.3 Supports and Restraints

1.3.1 Supports

Left Support	LS = Pinned
Right Support	RS = Pinned
Effective Buckling Length	L _b = 8 ft

1.4 Fabrication

Fabrication Type = **Cellular Beam**

1.4.1 Details

Spacing Width $e = 4.5$ in
Opening Diameter $D_o = 12.3$ in

1.4.2 Serviceability Load Combinations

No	Combination
SLS 1	DL + LL

1.4.3 Ultimate Load Combinations

No	Combination
ULS 1	1.2DL + 1.6LL

1.5 Loads and Combinations

1.5.1 Basic Load Cases

Collateral Loads (CL) **No**
Snow Loads (SL) **No**
No of Wind Cases $WL = 0$

1.5.2 Loads and Combinations

Load No	Load Case	Load Type	Intensity (kips/ft)	Start Location (ft)	End Location (ft)
L1	DL	Distributed	0.125	0	40
L2	LL	Distributed	0.1	0	40

Auto Load Combinations **No**

1.6 Material Properties

Steel Grade **A992-Gr50**
Minimum Yield Stress $F_y = 50$ ksi
Tensile Stress $F_u = 65$ ksi

1.7 Deflection Limits

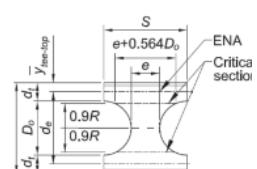
Live Load $DFL = 240$
Total Load $DFT = 180$

2.0 OUTPUT

ANSI/AISC 360-16

2.1 Section Details

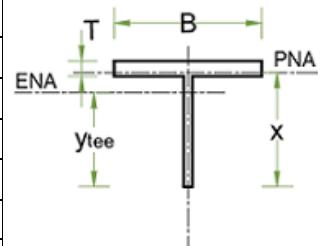
Spacing of Openings $S = e + D_o = 16.8$ in
Loss $= D_o / 2 - \sqrt{(D_o / 2)^2 - ((S - D_o) / 2)^2} = 0.43$ in



Depth of Expanded Beam	$d_g = d + D_o / 2 - \text{loss} = \mathbf{17.62 \text{ in}}$
Net Depth of Tee	$d_{t,\text{net}} = (d_g - D_o) / 2 = \mathbf{2.66 \text{ in}}$
	$y = \text{SQRT}((0.5 * D_o)^2 - (0.225 * D_o)^2) = \mathbf{5.49 \text{ in}}$
Critical Depth of Tee	$d_{t,\text{crit}} = D_o / 2 - y + d_{t,\text{net}} = \mathbf{3.32 \text{ in}}$

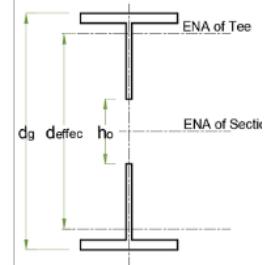
2.2 Top and Bottom Tee Section Properties

At Center of Opening		
$A_{\text{tee-net}} = \mathbf{1.38 \text{ in}^2}$	$x_{\text{net}} = \mathbf{2.53 \text{ in}}$	$r_x_{\text{net}} = \mathbf{0.76 \text{ in}}$
$r_y_{\text{net}} = \mathbf{0.92 \text{ in}}$	$y_{\text{tee-net}} = \mathbf{2.08 \text{ in}}$	$S_{x\text{top-net}} = \mathbf{1.38 \text{ in}^3}$
$S_{x\text{bot-net}} = \mathbf{0.39 \text{ in}^3}$	$Z_x_{\text{net}} = \mathbf{0.68 \text{ in}^3}$	$I_{x\text{tee-net}} = \mathbf{0.803 \text{ in}^4}$
$I_{y\text{tee-net}} = \mathbf{1.175 \text{ in}^4}$	$J_{\text{net}} = \mathbf{0.022 \text{ in}^4}$	$y_o_{\text{net}} = \mathbf{1.97 \text{ in}}$
At Critical Section		
$A_{\text{tee-crit}} = \mathbf{1.51 \text{ in}^2}$	$x_{\text{crit}} = \mathbf{3.16 \text{ in}}$	$r_x_{\text{crit}} = \mathbf{1.51 \text{ in}}$
$r_y_{\text{crit}} = \mathbf{1.18 \text{ in}}$	$y_{\text{tee-crit}} = \mathbf{2.53 \text{ in}}$	$S_{x\text{top-crit}} = \mathbf{1.9 \text{ in}^3}$
$S_{x\text{bot-crit}} = \mathbf{0.6 \text{ in}^3}$	$Z_x_{\text{crit}} = \mathbf{1.05 \text{ in}^3}$	$I_{x\text{tee-crit}} = \mathbf{1.505 \text{ in}^4}$
$I_{y\text{tee-crit}} = \mathbf{1.175 \text{ in}^4}$	$J_{\text{crit}} = \mathbf{0.024 \text{ in}^4}$	$y_o_{\text{crit}} = \mathbf{2.42 \text{ in}}$
A_{tee} Area of Tee, x : Plastic Neutral Axis, r_x Radius of Gyration - X Axis, r_y Radius of Gyration - Y Axis, S_{x_{top}} Elastic Section Modulus - Top, S_{x_{bot}} Elastic Section Modulus - Bottom, Z_x Plastic Section Modulus, I_{x_{tee}} Moment of Inertia - X Axis, I_{y_{tee}} Moment of Inertia - Y Axis, J Polar Moment of Inertia, y_o Shear Center from Centroid,		



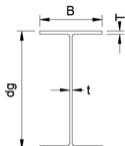
2.3 Net Section Properties

At Center of Opening		
$A_{\text{net}} = \mathbf{2.76 \text{ in}^2}$	$d_{\text{effecnet}} = \mathbf{16.46 \text{ in}}$	$I_{x\text{net}} = \mathbf{188.61 \text{ in}^4}$
$S_{x\text{net}} = \mathbf{21.404 \text{ in}^3}$	$Z_{x\text{net}} = \mathbf{22.724 \text{ in}^3}$	
At Critical Section		
$A_{\text{crit}} = \mathbf{3.02 \text{ in}^2}$	$d_{\text{effec-crit}} = \mathbf{16.04 \text{ in}}$	$I_{x\text{crit}} = \mathbf{197.536 \text{ in}^4}$
$S_{x\text{crit}} = \mathbf{22.417 \text{ in}^3}$	$Z_{x\text{crit}} = \mathbf{24.255 \text{ in}^3}$	



2.4 Gross Section Properties

Gross Area	$A_{\text{gross}} = A_{\text{net}} + D_o * t = \mathbf{5.22 \text{ in}^2}$
Gross Moment of Inertia	$I_{x\text{gross}} = I_{x\text{net}} + (t * D_o^3 / 12) = \mathbf{219.625 \text{ in}^4}$
Gross Elastic Section Modulus	$S_{x\text{gross}} = I_{x\text{gross}} / (D_g / 2) = \mathbf{24.924 \text{ in}^3}$



2.5 Capacity of Tee Section

2.5.1 Section Classification

Axial Compression (Table B4.1.a)				
Element	Slenderness Limit	Actual	Local Classification	Section Classification
Flange	$0.56(E_s/F_y)^{1/2} = \mathbf{13.5}$	$(B/2)/T = \mathbf{8.8}$	Non-Slender	Non-Slender
Tee Stem	$0.75(E_s/F_y)^{1/2} = \mathbf{18.1}$	$d_t/t = \mathbf{16.6}$	Non-Slender	

Flexure (Table B4.1.b)

Element (Table C4-1)	Compact Limit	Non-Compact Limit	Actual	Local Classification	Section Classification
Flange	$0.38(F_s/F_y)^{1/2} = 9.2$	$1.0(F_s/F_y)^{1/2} = 24.1$	$(B/2)/T = 8.8$	Compact	Compact
Element Tee Stem	Compact Limit $0.84(F_s/F_y)^{1/2} = 20.2$	Non-Compact Limit $1.03(F_s/F_y)^{1/2} = 36.6$	Actual $d_t/t = 16.6$	Local Classification Compact	Section Classification Compact

2.5.2 Axial Compression Strength of Tee

a. Flexural Buckling :

Note: Design Assumptions are per Clause 3.2.2.1 of AISC Steel Design Guide 31

Slenderness x - axis	$L_c/r_x = 0.65 * e / r_x = 4.01$	
Slenderness y - axis	$L_c/r_y = e / r_y = 6.98$	
Critical Slenderness	$L_c/r = 6.98$	
Elastic Buckling Stress	$F_e = \pi^2 * E_s / (L_c/r)^2 = 5881.4 \text{ ksi}$	<i>Eq. E3-4</i>
Limiting Elastic Slenderness	$(L_c/r)_l = 4.71 * (E_s / F_y)^{0.5} = 113.4$	
Critical Stress	$F_{cr1} = 0.658^{(F_y/F_e)} * F_y = 49.9 \text{ ksi}$	

b. Flexural Torsional Buckling :

Shear Modulus of Elasticity of Steel	$G = 11200.0 \text{ ksi}$	
Elastic Critical Buckling Stress about Minor Axis	$F_{ey} = (\pi^2 * E_s) / (e / r_y)^2 = 5881.4 \text{ ksi}$	
Polar Radius of Gyration about the Shear Center	$r_o = (\gamma_o^2 + ((I_{x\text{tee-crit}} + I_{y\text{tee}}) / A_{\text{tee-crit}}))^{0.5} = 2.8 \text{ in}$	<i>Eq. E4-9</i>
Elastic Critical Torsional Buckling Stress	$F_{ez} = ((\pi^2 * E_s / e^2) + G * J) * (1 / (A_{\text{tee-crit}} * r_o^2)) = 679.3 \text{ ksi}$	
Flexural Constant	$H_c = 1 - (\gamma_o^2 / r_o^2) = 0.2$	<i>Eq. E4-8</i>
Elastic Critical Buckling Stress	$F_{et} = (F_{ey} + F_{ez}) / (2.0 * H_c) * (1.0 - ((1.0 - (4.0 * F_{ey} * F_{ez}) * H_c / (F_{ey} + F_{ez})^2))^{0.5}) = 622.7 \text{ ksi}$	
Critical Stress	$F_{cr2} = 0.658^{(F_y/F_{et})} * F_y = 48.4 \text{ ksi}$	<i>Eq. E3-2</i>

c. Compressive Strength

Critical Stress	$F_{cr} = \min(F_{cr1}, F_{cr2}) = 48.4 \text{ ksi}$	
Effective Flange Width	$b_e = 3.97 \text{ in (Non-Slender)}$	<i>Eq. E7-1</i>
Effective Tee Stem Depth	$d_e = 3.32 \text{ in (Non-Slender)}$	<i>Eq. E7-1</i>
Effective Sectional Area	$A_e = b_e * T + (d_e - T) * t = 1.51 \text{ in}^2$	
Nominal Compressive Strength	$P_n = A_e * F_{cr} = 73.2 \text{ Kips}$	
Available Compressive Strength	$P_u = \varphi_c * P_n = 65.9 \text{ Kips}$	

2.5.3 Flexure Strength of Tee

a. Yielding

Yield Moment about Axis of Bending	$M_y = F_y * S_{x\text{bot-crit}} = 29.8 \text{ kips-in}$	<i>Eq. F9-3</i>
Plastic Bending Moment	$M_p = M_y = 29.8 \text{ kips-in}$	<i>Eq. F9-4</i>

b. Lateral-torsional Buckling

Lateral Torsional Buckling Factor	$B_{\text{tee}} = -2.3 * (d_{t\text{-net}} / L_b) * (I_{y\text{tee-net}} / J)^{0.5} = -0.5$	<i>Eq. F9-12</i>
Nominal Flexural Strength based on Lateral Torsional Buckling	$M_{cr} = 1.95 * E_s / L_b * (I_{y\text{tee-net}} * J)^{0.5} * (B_{\text{tee}} + (1 + B_{\text{tee}})^2)^{0.5} = 60.2 \text{ kips-in}$	<i>Eq. F9-10</i>

c. Local Buckling of Tee Stem

The Tee Stem is compact, local buckling of Tee Stem is not applicable.

d. Local Buckling of Tee Flange

The Tee Flange is compact, local buckling of Tee Flange is not applicable.

e. Flexural Strength:

Available Flexural Strength $M_a = \phi_b * \text{Min}(a, b, c, d) = \mathbf{26.8 \text{ kips-in}}$

2.6 Flexural Strength of Web Post

Note: The calculation of web post flexural strength is per Clause 3.4.2 of AISC Steel Design Guide 31

Elastic Moment $M_e = t * (S - D_o + 0.564 * D_o)^2 / 6.0 * F_y = \mathbf{218.0 \text{ kips-in}}$

Allowable / Elastic Moment Ratio $M_{al}/M_e = \mathbf{0.425}$

Available Flexural Strength $\phi M_n = \phi_b * (M_{al}/M_e) * M_e = \mathbf{83.3 \text{ kips-in}}$

2.7 Vertical Shear Capacity of Net Section

Depth to Thickness Ratio $h/t = d_{t,\text{net}} / t = \mathbf{13.3}$

Web plate buckling coefficient $k_v = \mathbf{1.2}$

Web Shear Buckling Coefficient $C_{v2} = \mathbf{1.0}$

Nominal Shear Strength $\phi_{vv\text{net}} = \phi_v * 0.6 * F_y * (2 * d_{t,\text{net}} * t) * C_{v2} = \mathbf{31.9 \text{ Kips}}$ Eq. G3-1

2.8 Horizontal Shear Capacity of Web Post

$\phi_v = \mathbf{1.0}$

Available Shear Strength $\phi_{vv\text{nhori}} = \phi_v * 0.6 * F_y * (e * t) = \mathbf{27.0 \text{ Kips}}$ Eq. J4-3

2.9 Vertical Shear Capacity of Beam Gross Section

Depth to Thickness Ratio $h/t = (d_g - 2 * T) / t = \mathbf{85.9}$

Web Plate Shear Buckling Coefficient $k_{v2} = \mathbf{5.3}$

Web Shear Strength Coefficient $C_{v1} = \mathbf{0.7}$

$\phi_{v2} = \mathbf{0.9}$

Nominal Shear Strength $\phi_{vv\text{ngross}} = \phi_{v2} * 0.6 * F_y * (d_g * t) * C_{v1} = \mathbf{67.8 \text{ Kips}}$ Eq. G2-1

2.10 Design Forces action at the opening and web post

Tee Axial Force $P_r = M_r / d_{\text{effec}}$

Horizontal Shear $V_h = P_{r(i+1)} - P_{ri}$

Vierendeel Moment $M_{vr} = V_r / 2 * (e / 2)$

Web Post $M_u = V_h * h$

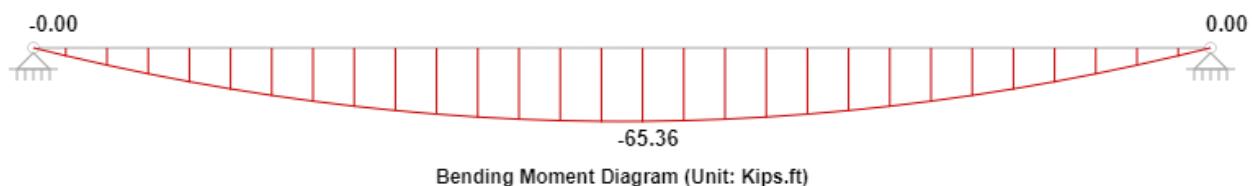
Axial Force and Moment Interaction

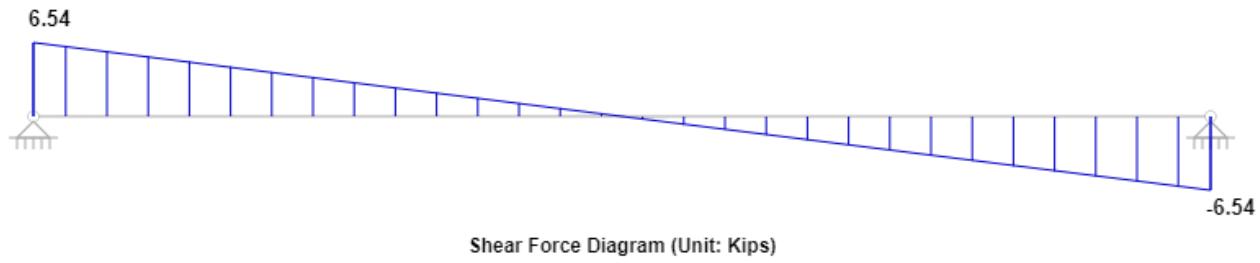
$$\text{For } P_r/P_c \geq 0.2 = (P_r / P_u) + (8 / 9) * (M_{vr} / M_a) \quad \text{Eq. H1-1a}$$

$$\text{For } P_r/P_c < 0.2 = (P_r / P_u) / 2 * (M_{vr} / M_a) \quad \text{Eq. H1-1b}$$

2.11 Beam Analysis and Design Forces

Combination: 1.2DL + 1.6LL

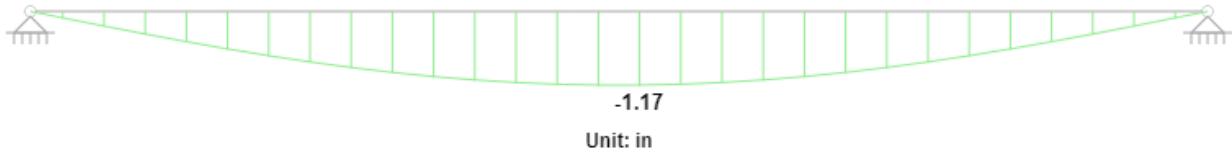




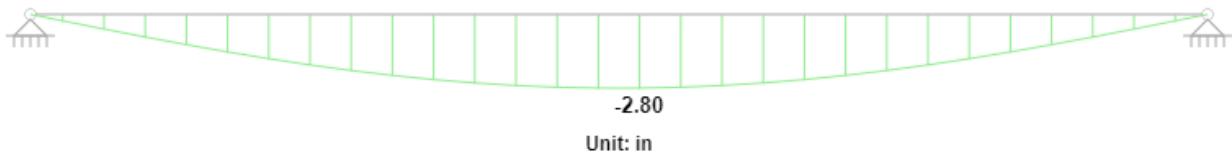
Opening	1	2	3	4	5	6	7	8
	Global Shear V_r	Global Moment M_r	Tee Axial P_r M_r/d_{eff}	Hr. Shear V_h $P_{r(i+1)} - P_{r(i)}$	Vir. Mom. M_{vr} $V_r/2 * (Do/4)$	Web Post M_u $V_h * h$	P_r/P_c	Interaction
1	6.25	5.64	4.22	6.29	9.60	34.84	0.064	0.390
2	5.79	14.05	10.51	5.82	8.90	32.23	0.160	0.412
3	5.33	21.84	16.34	5.34	8.20	29.58	0.248	0.520
4	4.87	28.98	21.68	4.86	7.49	26.93	0.329	0.578
5	4.42	35.48	26.55	4.39	6.79	24.27	0.403	0.628
6	3.96	41.35	30.93	3.91	6.09	21.62	0.470	0.671
7	3.50	46.57	34.84	3.43	5.38	18.97	0.529	0.707
8	3.04	51.15	38.27	2.95	4.68	16.32	0.581	0.736
9	2.59	55.09	41.21	2.47	3.98	13.66	0.626	0.757
10	2.13	58.39	43.68	1.99	3.27	11.01	0.663	0.772
11	1.67	61.05	45.67	1.51	2.57	8.36	0.693	0.778
12	1.21	63.07	47.18	1.03	1.87	5.71	0.716	0.778
13	0.76	64.44	48.21	0.55	1.16	3.05	0.732	0.770
14	0.30	65.18	48.77	0.11	0.46	0.62	0.740	0.755
15	0.08	65.33	48.88	0.21	0.12	1.14	0.742	0.746
16	0.39	65.06	48.67	0.65	0.60	3.58	0.739	0.759
17	0.85	64.19	48.03	1.13	1.30	6.23	0.729	0.772
18	1.30	62.69	46.90	1.60	2.00	8.88	0.712	0.778
19	1.76	60.54	45.30	2.08	2.71	11.53	0.688	0.777
20	2.22	57.76	43.21	2.56	3.41	14.19	0.656	0.769
21	2.68	54.33	40.65	3.04	4.11	16.84	0.617	0.753
22	3.13	50.27	37.61	3.52	4.82	19.49	0.571	0.731
23	3.59	45.56	34.09	4.00	5.52	22.14	0.517	0.701
24	4.05	40.21	30.09	4.48	6.22	24.80	0.457	0.663
25	4.51	34.23	25.61	4.96	6.93	27.45	0.389	0.618
26	4.96	27.60	20.65	5.44	7.63	30.10	0.313	0.567
27	5.42	20.33	15.21	5.92	8.33	32.75	0.231	0.507
28	5.88	12.42	9.29	5.92	9.04	32.75	0.141	0.408

2.12 Check Deflection

Live Load



Critical: DL + LL



Allowable Live Load Deflection

$$AD_{LL} = BL / DFL = 2.00 \text{ in}$$

Allowable Total Load Deflection

$$AD_{TL} = BL / DFT = 2.67 \text{ in}$$

3.0 SUMMARY

Description	Critical		Actual	Allowable	Status
	Comb.	Location (ft)			
Tee - Axial Force (Kips)	ULS 1	20.488	48.88	65.88	PASS
Tee - Moment (kips-in)	ULS 1	0.888	9.60	26.79	PASS
Tee - Compression + Moment Interaction	ULS 1	14.888	0.778	1.000	PASS
Web Post - Moment (kips-in)	ULS 1	0.888	34.84	83.32	PASS
Web Post - Horizontal Shear (Kips)	ULS 1	0.888	6.29	27.00	PASS
Net Section - Vertical Shear (Kips)	ULS 1	0.888	6.25	31.94	PASS
Deflection - Live Load (in)	LL	-	1.17	2.00	PASS
Deflection - Total Load (in)	SLS 1	-	2.80	2.67	FAIL

